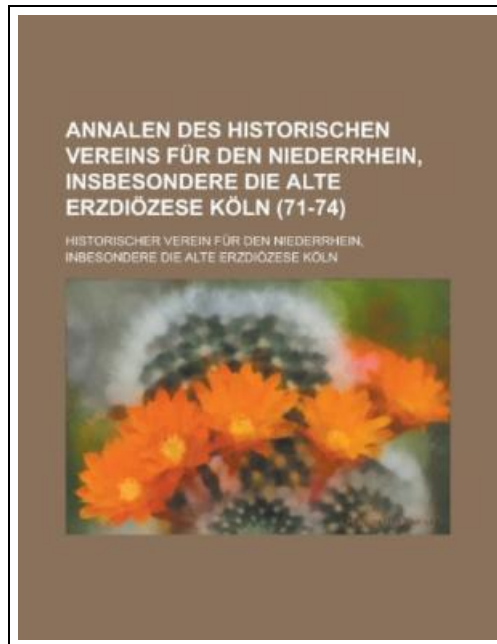


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ANNALEN DES HISTORISCHEN VEREINS FUR DEN NIEDERRHEIN, INSBESONDERE DIE ALTE ERZDIOEZE KOLN (71-74)



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RareBooksClub. Paperback. Book Condition: New. This item is printed on demand. Paperback. 74 pages. Original publisher: Torrance, Calif. : Science Applications International Corporation ; Houston, Tex. : National Aeronautics and Space Administration, Lyndon B. Johnson Space Center, 1990 OCLC Number: (OCoLC)245529706 Excerpt: . . . PayloadAccommodations TheRRSis to beflown in a heatshielddown launchconfiguration to maintain a constant axial acceleration load direction during the Sgmechrt entire mission. A payload fairing is required to protect the RRriottapa during ascent. The most important constraint imposed by the faisais is on the size of the RRS (i. e. , diameter and length). For dt ELVs a large enough fairing n cannot be developed. Payloa fitting (PAF) constraint on RRSc. g. were not assessed i of the contract. Payload Accessibility RRSclose-out is to occur upto T-4 hours before launch. Close-out, is assumed to involve human access (e. g. , final inspections detachmen of umbilicals, rodent cage replacement). Flight Environment Thermal loads during launch vary moderately among ELVs. tSome alser candidate ELVs may require fairing insulation to reduce internada temperatures below RRS tolerance. Acceleration and shock lore were also determined. It is required that the ELV not impose g than 10 g axial acceleration loads on the rodents. Cost per Flight Life cycle cost was a key issue regarding selection of the candidate ELV fleet. 5. 2 Maturity Evaluation Factors ELV maturity was considered a top-level programmatic concern which should be addressed after technical merit was established. Launch vehicle maturity factors were evaluated and included the following: Company Relevant Experience and Past Performance ELV Design Approach Development Status Recurring Cost Risk These factors were introduced to establish a confidence level in the proposed ELV availability and performance. It is desirable to choose a flight proven, operational launch system as opposed to a proposed design involving potential development, cost, and schedule risk. 6. 0 LAUNCH VEHICLE DATA Data for each of the candidate ELVs was obtained with r. . . This item ships from La Vergne, TN. Paperback.



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